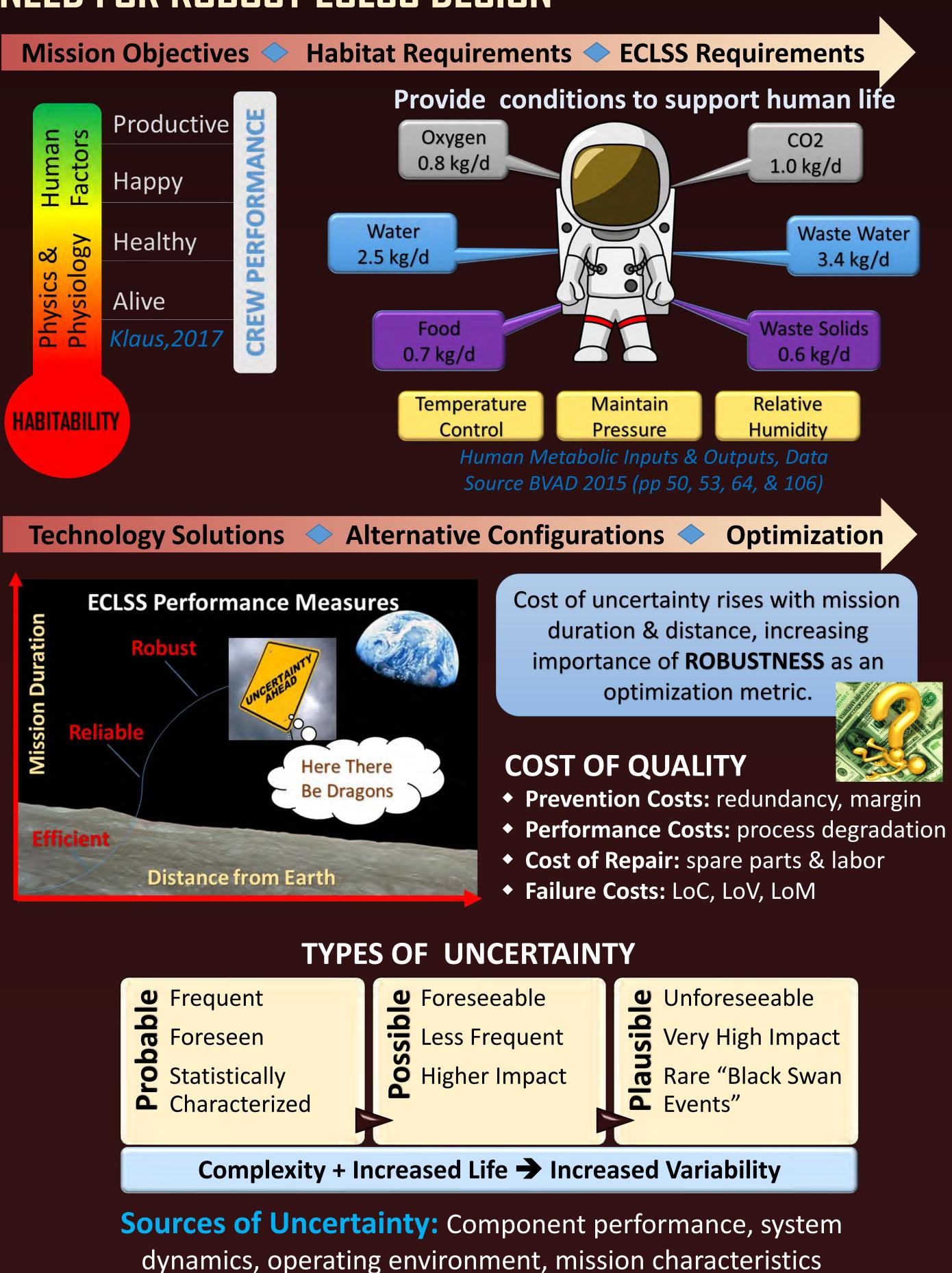
ABSTRACT

Life support system designs for human space exploration can include many different combinations of technologies. A variety of metrics might be used to determine the "best" configuration, such as efficiency (in mass, volume, or power), safety, reliability, and robustness. Mission characteristics will dictate the relative importance of these factors. For sustainable deep space exploration, as mission duration and distance from Earth increases, robustness may become the more important design criteria. The goals of this research are to define metrics and propose design practices for optimizing ECLSS robustness so that sustainable environmental control and life support systems can be realized for long term space missions.

RESEARCH OBJECTIVES

- Objective 1: Define a quantitative, measurable robustness metric for spacecraft ECLSS
- Objective 2: Provide design guidance for improving ECLSS robustness
- Objective 3: Demonstrate a methodology for assessing robustness of an ECLSS design

NEED FOR ROBUST ECLSS DESIGN



Quantifying ECLSS Robustness for Deep Space Exploration

Christine Escobar, Space Lab Technologies, LLC, Boulder, CO Dr. James A. Nabity, University of Colorado at Boulder

A ROBUST DESIGN METHODOLOGY FOR ECLSS

Robust: "Capable of performing without failure under a wide range of conditions"

Merriam-Webster "Often [spacecraft] systems are forced to operate under conditions which deviate significantly from ideal design conditions. A degree of how well a system performs with no appreciable degradation in performance under such conditions is measured by its robustness." Miller et al. (2008)

ECLSS robustness is its ability to maintain habitable conditions for crew survival and productivity over the mission lifetime under a wide range of conditions.

ECLSS robustness includes insensitivity of performance (i.e. maintaining habitability) to 1) Random expected failures and conditions (reliability) 2) Foreseen but unexpected deviations in conditions or disturbances (resilience)

3) Unforeseen disturbances or adverse events (survivability)

ROBUST SYSTEM CHARACTERSTICS



RELIABLE

RESILIENT

A new "robustness" metric is needed to describe system availability in off-nominal conditions, or abnormal use.

Evolution of Robust Design Methodology

Robust design practices **evolved from** quality engineering and industrial process control, starting with the ideas of Genichi Taguchi in Japan.

The engineering community generally agrees that variation in usage conditions or inputs imparts quality loss, and that the goal of robust design is to find control factors (i.e. design features) that reduce sensitivity to noise.

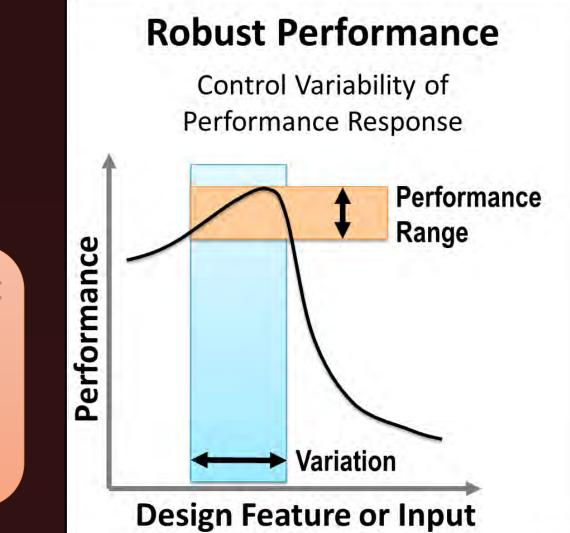
General Robust Design Methodology	Meth
1. Define key product characteristic (KPC):	Define
2. Identify & characterize variation sources:	Charac condit
3. Define or model system behavior:	Mathe
4. Quantify robustness of KPC given variation & system model:	Need a
5. Select or improve design:	Identif habita

Escobar et al., 2017





SURVIVABLE



nodology for Robust ECLSS Design e "Habitability"

acterize ECLSS inputs, operating tions, component reliability, etc.

ematical or physical ECLSS model

an ECLSS robustness metric

fy design features contributing to habitability loss w/ minimum cost of quality

QUANTIFYING ECLSS ROBUTNESS Habitability is the ECLSS Key Product Characteristic (KPC)

Potential Habitability Contributors (y_i) Apply to Utility Functions, $H_i \in [0,1]$ **Contributors include O₂ partial pressure, CO₂** partial pressure, total cabin pressure, wet bulb temperature, food & water availability and quality, or even the presence of noxious substances.

Habitability Index Definition

- 1. *H* must be 1 when crew performance
- capacity is full \rightarrow all H_i are equal to 1.
- 2. H must be 0 under any fatal conditions, i.e. when any $H_i = 0$.
- 3. H must be no better than any
- individual H*i*, i.e. $H \le \min(H_i)$. 4. The impact of H_i on H is not independent. A reduction in one H_i increases the impact of another H_i .

$H = \Pi H i$, for $i = 1,...,n \& H_i \in [0,1]$

1. Variance

- **2. Effective Fitness** E(y)
- 3. Minimax Optimization (Worst Case Philosophy)
- **Process Capability Index**

"Habitability Loss" Let $L_{H} = (H-1)^{2}$

> **ECLSS Robustness**: $R_{H} = 1 - \sqrt{E(L_{H})} = 1 - \sqrt{[(1 - E(H))^{2} + Var(H)]}$ 'spread'

IMPROVING ECLSS ROBUSTNESS

Minimize Cost Of Quality

Robustness Normalized Equivalent System Mass (ESM):

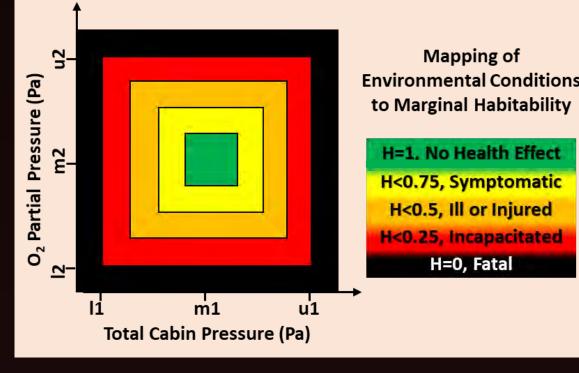
Equivalent mass required to achieve equivalent robustness

 $ESM_{R} = ESM/R_{H}$

REFERENCES

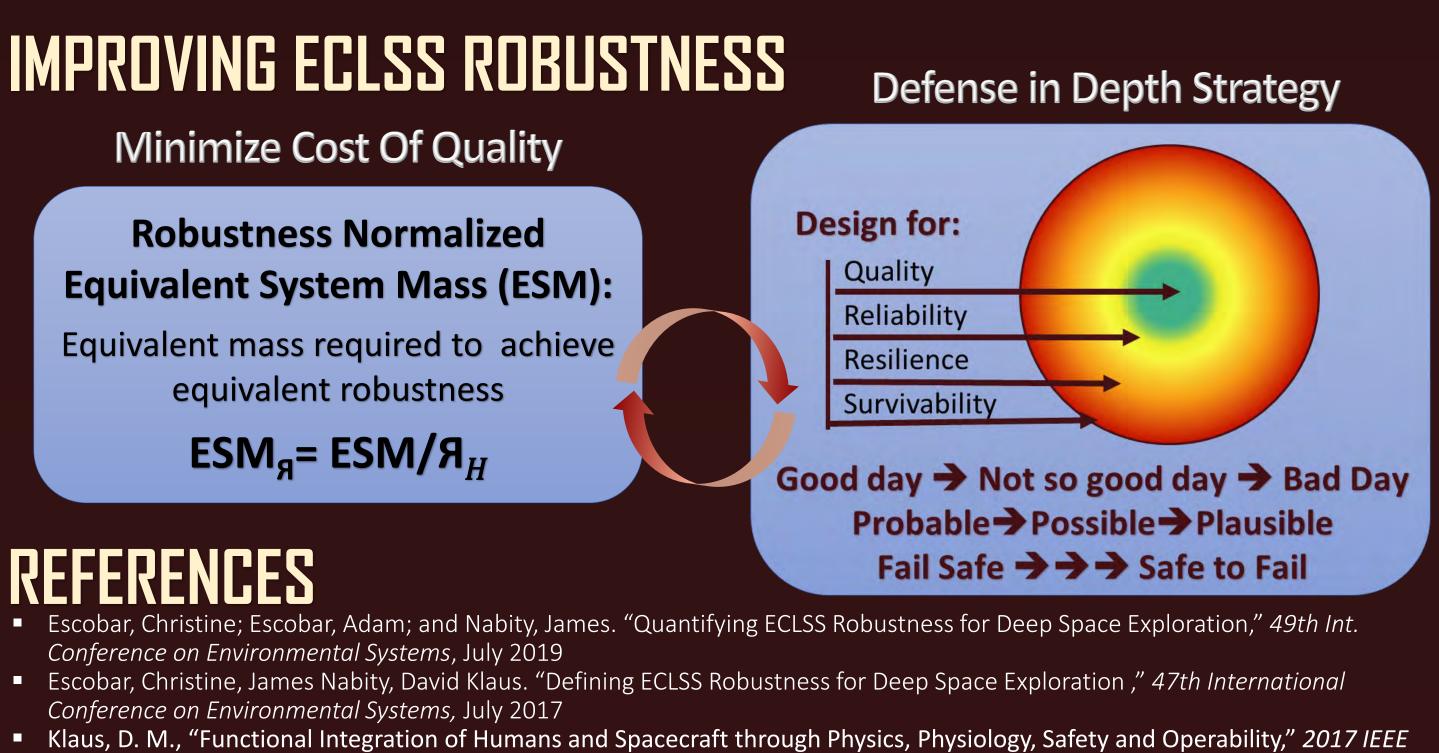
- Conference on Environmental Systems, July 2019
- Conference on Environmental Systems, July 2017
- Aerospace Conference, 2017
- Human Rated Spacecraft Systems," NASA/TM-2008-215126/Vol II, 2008.





→ Habitability Over Mission Duration 'Reliability 'Resilience' $\sim\sim\sim\sim\sim$ 2 'Survivability Syste Time of Disturbance

Many Possible Robustness Metrics 5. Quality Loss 9. Variation Risk Priority # **6. Sensitivity** $(\delta y / \delta x)$ **10.Information Content** (Axiomatic Design) 7. Signal to Noise (Taguchi) 8. Mean & Variance, See Escobar et al., **2019 for details** Weighted Sum **Taguchi's "Quality Loss" Function Accounts for Bias & Spread:** Min $E(L) = E[k(y-m)^2] = k[(\mu-m)^2 + \sigma^2],$ where *m* is target value of KPC & k is a cost factor **Expected Habitability Loss** $E[L_{H}] = E[(H-1)^{2}] = [1-E(H)]^{2} + Var(H)^{-1}$



• Miller, J., Leggett, J., and Kramer-White, J, "Design Development Test and Evaluation (DDTE) Considerations for Safe and Reliable